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INHYD: Computer Code for Intraply Hybrid Composite Design

Users Manual

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INHYD: Computer Code for Intraply Hybrid Composite Design

Users Manual

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Scientific and Technical Information Branch

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Summary

A computer program called INHYD has been developed for intraply hybrid composite design. This report is a users manual for INHYD. INHYD embodies several composite micromechanics theories, intraply hybrid composite theories, and an integrated hygrothermomechanical theory. INHYD can be run in both interactive and batch modes. It has considerable flexibility and capability, which the user can exercise through several options. These options are demonstrated through appropriate INHYD runs in the manual.

Introduction

Intraply hybrid composites have been investigated theoretically and experimentally at Lewis over the past several years. Theories developed during these investigations and corroborated by attendant experiments were used to develop a computer program identified as INHYD (INtraply HYbrid composite Design). INHYD was designed to operate in the interactive and batch modes. This report is a users manual for INHYD.

INHYD includes several composite micromechanics theories, intraply hybrid composite theories, and an integrated hygrothermomechanical theory. Equations from these theories are used by the program as appropriate for the user's specific applications. The elements and subroutines in INHYD are summarized in appendix A. The capabilities of the INHYD computer code are described in appendix B. The symbols used in the program are defined in appendix C. Before running INHYD for the first time, the user should read the program capability description in appendix B, which is a reproduction of NASA TM-82593 included herein for completeness.

First, the manual presents examples for the interactive mode—for a dry, room-temperature case and then for a case with moisture present. Then an interactive case is described, where the data are preentered as they would be for a batch case. Next batch cases are described. After following through the test cases given, a user should be able to use INHYD either in the interactive or batch mode and with or without temperature and moisture effects to predict properties for uniaxial intraply hybrid composites.

INHYD Interactive Mode

Ambient Conditions, Dry

To run INHYD, the user must first install and compile the program on his/her computer according to the system in use. The procedure used on the Lewis Research Center's UNIVAC 1100 is described in detail here by using the actual computer printout. Computer prompt signals are identified with upper-case letters. User entries follow the computer symbol >. Appropriate explanatory notes follow each printout.

```
>@ass,a inhyd.

READY
>@map,i .abs

MAP 30R1(C1) S74T11 04/22/83 11:12:46 (->0)
>in inhyd.

>end

START=012516, PROG SIZE(I/D)=10822/4956
SYS**RLIB*. LEVEL
END MAP. ERRORS: 0
>@xat .abs

INHYD TO BE RUN BATCH OR INTERACTIVE (B/I)
>i
```

As illustrated, the user is prompted by the computer, and his/her responses or entries are the input required for the run.

```
INHYD INTERACTIVE DATA ENTRY
SECONDARY COMPOSITE VOLUME RATIO
>.2
```

The user selected 0.2 for the secondary composite volume ratio.

```
WHICH COMPOSITE DATA TO BE ENTERED (P/S)
```

The choice was primary (P) or secondary (S); the user selected primary.

```
PROPERTIES TO BE ENTERED (C/F/M/T) >f
```

The choice was composite (C), fiber (F), dry matrix (M), or matrix with moisture and temperature effects (T); the user selected fiber properties for the primary composite.

```
INPUT DATA HEADING (1 TO 80 CHARACTERS)
```

This heading identifies the primary fiber property table to be printed out by the computer.

```
ENTER THE FIBER PROPERTY VALUES
        EFF2
               GFP12 GFP23
                                NUFP12 NUFP23
>.32e8,.2e7,.2e7,.1e7,.2,.25
CTEFP1 CTEFP2 RHOFP
                        NEP
                                DIFF
                                        CEEC
>-.56e-6,.56e-5,.63e-1,.1e5,.3e-3,.17
 KFP1
        KFP2
                KFP3
                        SFPT
                                SEPC
>.58e3,.58e2,.58e2,.4e6,.4e6
```

Here the user has entered the 17 values for the primary fiber properties that will be used in the micromechanics for calculating the primary composite properties. (Entry headings are defined in appendix C.)

```
PROPERTIES TO BE ENTERED (C/F/M/T)
```

Dry matrix (M) was selected.

```
INPUT DATA HEADING (1 TO 80 CHARACTERS)
```

This heading identifies the primary matrix property table to be printed out by the computer.

```
ENTER THE MATRIX PROPERTY VALUES
EMF GMF NUMF CTEMP RHOMF CMPC
>.46e6,.1643e6,.4,.32e-4,.443e-1,.25

KMP SMPT SMPC SMPS BTAMP DIFMF
>.125e1,.68e4,.363e5,.7e4,.4,.2e-3
```

Here the user has entered the 12 values for the primary matrix properties that will be used in the micromechanics for calculating the primary composite properties. (Entry headings are defined in appendix C.)

```
COMPOSITE HEADING (1 TO 80 CHARACTERS)
```

This heading identifies the primary composite property table. Table entries are to be calculated from the primary fiber properties and primary matrix properties and tabulated under this heading. The primary composite is AS/E (graphite/epoxy).

```
ENTER CONSTITUENT VOLUME RATIOS FIBER MATRIX VOID >.55,.44,.01
```

The fiber volume ratio, matrix volume ratio, and void volume ratio of the primary composite are entered as desired by the user. The sum of the three values must be 1.0.

```
VOID CONDUCTIVITY
```

The thermal conductivity of the voids was chosen as 0.1.

```
METHOD FOR CALCULATING COMPOSITE PROPERTIES (H/M)
```

The user can choose hygrothermomechanical theory (H) or micromechanics theory (M) for calculating the primary composite properties. Micromechanics theory was selected since moisture and temperature effects are not being studied.

```
WHICH COMPOSITE DATA TO BE ENTERED (P/S)
```

PROPERTIES TO BE ENTERED (C/F/M/T)

Now the secondary composite (S) properties will be calculated from the fiber and matrix properties about to be entered below. Since the procedure is the same as that used for the primary composite, fewer comments will be required.

```
INPUT DATA HEADING (1 TO 80 CHARACTERS)
>s-glass
The secondary fiber is S-Glass.
 ENTER THE FIBER PROPERTY VALUES
       EFS2
                GFS12 GFS23
                                NUFS12
EFS1
                                        NUFS23
>.124e8,.124e8,.517e7,.517e7,.2,.2
CTEFS1 CTEFS2 RHOFS
                        NFS
                                DIFS
                                         CESC
>.28e-5,.28e-5,.9e-1,.204e3,.36e-3,.17
        KFS2
                KFS3
                        SFST
>.75e1,.75e1,.75e1,.36e6,.3e6
PROPERTIES TO BE ENTERED (C/F/M/T)
 INPUT DATA HEADING (1 TO 80 CHARACTERS)
>3501-5,dry,70 desrees f
 ENTER THE MATRIX PROPERTY VALUES
                                RHOMS
 FMS
        GMS
                NUMS
                      CTEMS
                                         CMSC
>.46e6,.1643e6,.4,.32e-4,.443e-1,.25
 KMS
        SMST
                SMSC
                        SMSS
                                BTAMS
                                         DIFMS
>.125e1,.68e4,.363e5,.7e4,.4,.2e-3
```

The same matrix is used for the secondary composite as was used for the primary one. If desired, a different matrix could be used.

```
COMPOSITE HEADING (1 TO 80 CHARACTERS)
>s-glass/e

ENTER CONSTITUENT VOLUME RATIOS
FIBER MATRIX VOID
>.55,.44,.01

VOID CONDUCTIVITY
>.1

METHOD FOR CALCULATING COMPOSITE PROPERTIES (H/M)
>m

HYBRID HEADING (1 TO 80 CHARACTERS)
>as/e//e-glass/e,80/20
```

This is the heading for the hybrid composite table that will be calculated.

```
METHOD FOR CALCULATING HYBRID PROPERTIES (R/M)
```

The choice is micromechanics theory (M) or rule of mixtures (R); the user selected the rule of mixtures.

The computer then lists the headings and properties previously entered by the user for the fibers (17 properties) and matrix (12 properties) of the primary composite in tabular form. Next it lists the respective volume ratios and the 37 calculated properties for the primary composite and a table of flexural strengths for the primary composite as calculated by several theories.

PRIMARY FIBER PROPERTIES; AS GRAPHITE

1	ELASTIC MODULI	EFP1	.3200+08
2		EFP2	.2000+07
3	SHEAR MODULI	GFP12	.2000+07
4		GFP23	.1000+07
5	POISSON'S RATIO	NUFP12	.2000+00
6		NUFF23	.2500+00
7	THERM. EXP. COEF.	CTEFP1	5600-06
8		CTEFP2	.5600-05
9	DENSITY	RHOFP	.6300-01
10	NO. OF FIBERS/END	NFP	.1000+05
11	FIBER DIAMETER	DIFP	.3000-03
12	HEAT CAPACITY	CFPC	.1700+00
13	HEAT CONDUCTIVITY	KFP1	.5800+03
14		KFP2	.5800+02
15		KFP3	.5800+02
16	STRENGTHS	SFPT	.4000+06
17		SFPC	.4000+06

PRIMARY MATRIX PROPERTIES; 3501-5,DRY,70 DEGREES F

1	ELASTIC MODULUS	EMP	.4600+06
2	SHEAR MODULUS	GMP	.1643+06
3	POISSON'S RATIO	NUMP	.4000+00
4	THERM. EXP. COEF.	CTEMP	.3200-04
5	DENSITY	RHOMP	.4430-01
6	HEAT CAPACITY	CMPC	.2500+00
7	HEAT CONDUCTIVITY	KMP	.1250+01
8	STRENGTHS	SMPT	.6800+04
9		SMPC	.3630+05
10		SMPS	.7000+04
11	MOISTURE COEF	BTAMP	.4000+00
12	DIFFUSIVITY	DIFMP	.2000-03

PRIMARY COMPOSITE PROPERTIES; AS/E

BASED ON MIGROMECHANICS OF INTRAPLY HYBRID COMPOSITES: ELASTIC AND THERMAL PROPERTIES.

1	ELASTIC MODULI	EPC1	.1784+08
2		EPC2	.1187+07
2 3		EPC3	.1187+07
4	SHEAR MODULI	GPC12	.7087+06
5		GPC23	.5612+06
6		GPC13	.7087+06
7	POISSON'S RATIO	NUPC12	.2900+00
8		NUPC23	.3175+00
9		NUPC13	.2900+00
10	THERM. EXP. COEF.	CTEPC1	1280-06
11		CTEPC2	.1563-04
12		CTEPC3	.1563-04
13	DENSITY	RHOPC	.5414-01
14	HEAT CAPACITY	CPC	.1988+00
15	HEAT CONDUCTIVITY	KPC1	.3195+03
16		KPC2	.3702+01
17		KPC3	.3702+01
18	STRENGTHS	SPC1T	.2230+06
19		SPC1C	.2230+06
20		SPC2T	.4944+04
21		SPC2C	.2639+05
22		SPC12	.5335+04
23	MOIST. DIFFUSIVITY	DPC1	.8800-04
24		DPC2	.5168-04
25		DPC3	.5168-04
26	MOIST, EXP, COEF,	BTAPC1	.5402-02
27		BTAPC2	.1441+00
28		BTAPC3	.1441+00
29	FLEXURAL MODULI	EPC1F	.1784+08
30		EPC2F	.1187+07
31	STRENGTHS	SPC23	.5868+04
32		SPC1F	.2788+06
33		SPC2F	.1041+05
34		SPCSB	.8002+04
35	PLY THICKNESS	TPC	.5000-02
36	INTERPLY THICKNESS	PLPC	.5850-04
37	INTERFIBER SPACING	PLPCS	.5850-04

I*************************************	**************************************	I*************************************
I*************************************	I LONGITUDINAL	I TRANSVERSE I
I BILINEAR I SOLUTION	I .2230+06	I .8328+04 I
I*************************************	I	I ,9737+04 I
I*************************************		I .9737+04 I
I*************************************	1 .2788+06 I	I .1041+05 I
I*************************************	**************************************	**************************************

The longitudinal and transverse flexural strengths listed in the composite property table (items 32 and 33) are calculated by the parabolic theory. The ply thickness (item 35) is internally set at 0.005 in., which is a typical value for most fiber composites. The ply thickness is used to calculate flexural strength.

The computer repeats this procedure for the secondary composite—first listing the data entered by the user and then tabulating the 37 calculated properties for the secondary composite (which in this example is S-Glass/epoxy) and the flexural strengths.

SECONDARY FIBER PROPERTIES; S-GLASS

1	ELASTIC MODULI	EFS1	.1240+08
2		EFS2	.1240+08
3	SHEAR MODULI	GFS12	.5170+07
4		GFS23	. 5170+07
5	POÍSSON'S RATIO	NUFS12	.2000+00
6		NUFS23	.2000+00
7	THERM. EXP. COEF.	CTEFS1	.2800-05
8		CTEFS2	.2800-05
9	DENSITY	RHOFS	.9000-01
10	NO. OF FIBERS/END	NFS	.2040+03
11	FIBER DIAMETER	DIFS	.3600-03
12	HEAT CAPACITY	CFŚC	.1700+00
13	HEAT CONDUCTIVITY	KFS1	.7500+01
14		KFS2	.7500+01
15		KFS3	.7500+01
16	STRENGTHS	SFST	.3600+06
17		SFSC	.3000+06

SECONDARY MATRIX PROPERTIES; 3501-5, DRY, 70 DEGREES F

1	ELASTIC MODULUS	EMS	.4600+06
2	SHEAR MODULUS	GMS	.1643+06
3	POISSON'S RATIO	NUMS	.4000+00
4	THERM. EXP. COEF.	CTEMS	.3200-04
5	DENSITY	RHOMS	.4430-01
6	HEAT CAPACITY	CMSC	.2500+00
7	HEAT CONDUCTIVITY	KMS	.1250+01
8	STRENGTHS	SMST	.6800+04
9		SMSC	.3630+05
10		SMSS	.7000+04
11	MOISTURE COEF	BTAMS	.4000+00
12	DIFFUSIVITY	DIFMS	.2000-03

SECONDARY COMPOSITE PROPERTIES; S-GLASS/E

BASED ON MICROMECHANICS OF INTRAPLY HYBRID COMPOSITES: ELASTIC AND THERMAL PROPERTIES.

FIBER VOLUME RATIO - .550 MATRIX VOLUME RATIO - .440 VOID CONDUCTIVITY - .1000+00 VOID VOLUME RATIO - .010

1 2	ELASTIC MODULI	ESC1	.7061+07
		ESC2	.1881+07
3		ESC3	.1881+07
4	SHEAR MODULI	GSC12	.8449+06
5		GSC23	.8449+06
5 6		GSC13	.8449+06
7	POISSON'S RATIO	NUSC12	•2900+00
8		NUSC23	.2900+00
9		NUSC13	.2900+00
10	THERM. EXP. COEF.	CTESC1	.3779-05
11		CTESC2	.1348-04
12		CTESC3	.1348-04
13	DENSITY	RHOSC	.6899-01
14	HEAT CAPACITY	CSC	.1926+00
15	HEAT CONDUCTIVITY	KSC1	.4675+01
16		KSC2	.2750+01
17		KSC3	.2750+01
18	STRENGTHS	SSC1T	.2050+06
19		SSC1C	.1708+06
20		SSC2T	.4223+04
21		SSC2C	.2254+05
22		SSC12	.4842+04
23	MOIST. DIFFUSIVITY	DSC1	.8800-04
24		DSC2	.5168-04
25		DSC3	.5168-04
26	MOIST, EXP. COEF.	BTASC1	.1365-01
27		BTASC2	.1433+00
28		BTASC3	.1433+00
29	FLEXURAL MODULI	ESC1F	.7061+07
30		ESC2F	.1881+07

31	STRENGTHS	SSC23	.4842+04
32		SSC1F	.2329+06
33		SSC2F	.8892+04
34		SSCSB	.7263+04
35	PLY THICKNESS	TSC	.5000-02
36	INTERPLY THICKNESS	PLSC	.7020-04
37	INTERFIBER SPACING	PLSCS	.7020-04

I*************************************	**************************************	I*************************************
	LONGITUDINAL	TRANSVERSE I
I BILINEAR I	1864+06	.7114+04 I
I LINEAR RECTANGULAR I SOLUTION	.2019+06	8317+04 I
I************************ I NEUTRAL AXIS SHIFT I SOLUTION	: .2019+06 :	.8317+04 I
I*************************************	**************************************	**************************************
I RECTANGULAR I SOLUTION I***********************************	**************************************	**************************************

Finally the calculated hybrid composite property (35 properties) and hybrid composite flexural strength tables are printed out.

HYBRID COMPOSITE PROPERTIES; AS/E//A S-GLASS/E,80/20

BASED ON PREDICTION OF PROPERTIES OF INTRAPLY HYBRID COMPOSITES (RULE OF MIXTURES).

PRIMARY COMPOSITE VOLUME RATIO - .800
SECONDARY COMPOSITE VOLUME RATIO - .200

1	ELASTIC MODULI	EHC1	~1568+08
2		EHC2	.1326+07
3		EHC3	·1326+07
4	SHEAR MODULI	GHC12	.7360+06
5		GHC23	.6180+06
6		GHC13	.7360+06
6 7	FOISSON'S RATIO	NUHC12	.2900+00
8		NUHC23	.3120+00
9		NUHC13	.2900+00
10	THERM. EXP. COEF.	CTEHC1	.6534-06
11		CTEHC2	41520-04
12		CTEHC3	.1520-04
13	DENSITY	RHOHC	.5711-01
14	HEAT CAPACITY	CHC	.1976+00
15	HEAT CONDUCTIVITY	KHC1	.2566+03
16		KHC2	.3511+01
17		KHC3	.3511+01
18	STRENGTHS	SHC1T	.2194+06
19		SHC1C	.2126+06
20		SHC2T	.4800+04
21		SHC2C	.2562+05
22		SHC12	.5236+04
23	MOIST. DIFFUSIVITY	DHC1	.8800-04
24		DHC2	.5168-04
25		DHC3	.5168-04
26	MOIST. EXP. COEF.	BTAHC1	⊷ 7052-02
27		BTAHC2	1440+00
28		BTAHC3	.1440+00
29	FLEXURAL MODULI	EHC1F	.1568+08

30		EHC2F	.1326+07
31	STRENGTHS	SHC23	.5663+04
32		SHC1F	.2699+06
33		SHC2F	.1011+05
34		SHCSB	.7854+04
35	FIRER UNL. RATTO	UFH	.5500+00

I*************************************	**************************************	I*************************************
I*************************************	**************************************	*********************I I TRANSUERSE I **********
I BILINEAR I SOLUTION TXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXX	I .2159+06	I .8085+04 I
I LINEAR RECTANGULAR I SOLUTION	I .2193+06	I .9453+04 I
I SOLUTION	**************************************	****************************I I .9453+04 I I
I*************************************	**************************************	1 .1011+05 I
I*************************************	**************************************	*********************I I
1 ****************	(********************	*************

With Moisture Effects

An example for the same composite, 80/20 AS/E//S-G/E, but with moisture and temperature effects is presented below. Only features differing from those explained in the previous "dry" example are explained.

```
>@ass,a inhyd.
READY
>@map,i .abs
MAP 30R1(C1) S74T11 04/22/83 11:37:55 (->0)
>in inhed.
>end
START=012516, PROG SIZE(I/D)=10822/4956
SYS$*RLIB$. LEVEL
END MAP. ERRORS: 0 >@xqt .abs
 INHYD TO BE RUN BATCH OR INTERACTIVE (B/I)
           INHYD INTERACTIVE DATA ENTRY
SECONDARY COMPOSITE VOLUME RATIO
WHICH COMPOSITE DATA TO BE ENTERED (F/S)
PROPERTIES TO BE ENTERED (C/F/M/T)
INPUT DATA HEADING (1 TO 80 CHARACTERS)
>as graphite
```

```
ENTER THE FIBER PROPERTY VALUES
EFP1
      EFP2 GFP12 GFP23
                               NUFP12 NUFP23
>.32e8,.2e7,.2e7,.1e7,.2,.25
CTEFP1 CTEFP2 RHOFP NFP
                               DIFP
                                       CFPC
>-.56e-6,.56e-5,.63e-1,.1e5,.3e-3,.17
               KFP3
                               SFPC
KEP1
       KEP2
                       SEPT
>.58e3,.58e2,.58e2,.4e6,.4e6
PROPERTIES TO BE ENTERED (C/F/M/T)
```

Τ

The matrix with moisture and temperature properties (T) was selected. Now additional input is needed for the matrix. Nineteen entries are needed instead of the 12 required for the previous case:

```
INPUT DATA HEADING (1 TO 80 CHARACTERS)
>3501-5, with 2% moisture
 ENTER THE MATRIX PROPERTY VALUES
                       CTEMP RHOMP
                                        CMPC
 EMP
        GMP NUMP
>.46e6,.1643e6,.4,.32e-4,.443e-1,.25
        SMPT
                SMPC
                        SMPS
                                BTAMP
                                        DIFMP
>.125e1,.68e4,.363e5,.7e4,.4,.2e-3
 ENTER THE THERMAL/MOISTURE PROPERTY VALUES
        TGDR
                TEMP
                        MOIST (%)
>273.,460.,294.,2.0
```

The first three new entries in the above line are the reference temperature, the glass transition temperature of the resin, and the use temperature. Here they are entered in Kelvin, but any scale can be used since it is a ratio of differences. If Fahrenheit were used, the entries would be 32°, 368°, and 70°, respectively. The last entry is for 2 percent moisture. In addition the following three entries are needed:

```
KMST CMST RHOMST >.1,.1,1.0
```

These entries are the conductivity, the heat capacity, and the density of the moisture substance (see INHYD.INSTR).

```
COMPOSITE HEADING (1 TO 80 CHARACTERS)

>as/e with 2% moisture in matrix

ENTER CONSTITUENT VOLUME RATIOS

FIBER MATRIX VOID

>.55,.44,.01

VOID CONDUCTIVITY

>.1

METHOD FOR CALCULATING COMPOSITE PROPERTIES (H/M)
>h
```

Hygrothermomechanical theory was selected since 2 percent moisture is present in the matrix.

```
WHICH COMPOSITE DATA TO BE ENTERED (P/S) >s

PROPERTIES TO BE ENTERED (C/F/M/T) >f
```

```
INPUT DATA HEADING (1 TO 80 CHARACTERS)
 >5-41255
 ENTER THE FIBER PROPERTY VALUES
 EFS1 EFS2 GFS12 GFS23 NUFS12 NUFS23
>.124e8,.124e8,.517e7,.517e7,.2,.2
 CTEFS1 CTEFS2 RHOFS NFS
                                         CFSC
                                 DIFS
>.28e-5,.28e-5,.9e-1,.204e3,.36e-3,.17
       KFS2
               KFS3
                        SFST
->.75e1,.75e1,.75e1,.36e6,.3e6
 PROPERTIES TO BE ENTERED (C/F/M/T)
 INPUT DATA HEADING (1 TO 80 CHARACTERS)
>3501-5 with 2% moisture
 ENTER THE MATRIX PROPERTY VALUES
 EMS GMS NUMS CTEMS RHOMS
                                         CMSC
>.46e6,.1643e6,.4,.32e-4,.443e-1,.25
        SMST
                SMSC
                        SMSS
                               BTAMS
                                         DIEMS
>.125e1,.68e4,.363e5,.7e4,.4,.2e-3
 ENTER THE THERMAL/MOISTURE PROPERTY VALUES
TO TGDR TEMP
>273.,460.,294.,2.0
                         MOIST (%)
 KMST
        CMST
                 RHOMST
>.1,.1,1.0
 COMPOSITE HEADING (1 TO 80 CHARACTERS)
>s-glass with 2% moisture in matrix
 ENTER CONSTITUENT VOLUME RATIOS
FIBER MATRIX VOID
>.55,.44,.01
 VOID CONDUCTIVITY
METHOD FOR CALCULATING COMPOSITE PROPERTIES (H/M)
>h
 HYBRID HEADING (1 TO 80 CHARACTERS)
>as/e//s-g/e,80/20 with 2% moisture in matrix
METHOD FOR CALCULATING HYBRID PROPERTIES (R/M)
The choice is rule of mixtures (R) or micromechanical theory (M); the user selected the rule of
mixtures.
PRIMARY FIBER PROPERTIES; AS GRAPHITE
                                               .3200+08
            ELASTIC MODULI
                                   EFP1
                                               .2000+07
      2
                                   EFP2
                                               .2000+07
      3
            SHEAR MODULI
                                   GFP12
                                               .1000+07
                                   GFP23
            POISSON'S RATIO
      5
                                   NUFP12
                                              .2000+00
```

NUFP23

CTEFP1

THERM. EXP. COEF.

.2500+00

-.5600-06

8		CTEFP2	.5600-05
9	DENSITY	RHOFP	.6300-01
10	NO. OF FIBERS/END	NFP	.1000+05
11	FIBER DIAMETER	DIFP	.3000-03
12	HEAT CAPACITY	CFPC	.1700+00
13	HEAT CONDUCTIVITY	KFP1	.5800+03
14		KFP2	.5800+02
15		KFP3	.5800+02
16	STRENGTHS	SFPT	.4000+06
17		SFPC	.4000+06

PRIMARY MATRIX PROPERTIES; 3501-5, WITH 2% MOISTURE

REFERENCE TEMP. - 273.00
TEST TEMP. - 294.00
MOISTURE CONDUCTIVITY - .1000+00
MOISTURE DENSITY - .1000+01
DRY GLASS TRANS. TEMP. - 460.00
PCT. MOISTURE - 2.000
MOISTURE HEAT CAPACITY - .1000+00

ORIGINAL MATRIX PROPERTIES

1	ELASTIC MODULUS	EMP	.4600+06
2	SHEAR MODULUS	GMP	.1643+06
3	POISSON'S RATIO	NUMP	.4000+00
4	THERM. EXP. COEF.	CTEMP	.3200-04
5	DENSITY	RHOMP	.4430-01
6	HEAT CAPACITY	CMPC	.2500+00
7	HEAT CONDUCTIVITY	KMP	.1250+01
8	STRENGTHS	SMPT	.6800+04
9		SMPC	.3630+05
10		SMPS	.7000+04
11	MOISTURE COEF	BTAMP	.4000+00
12	DIFFUSIVITY	DIFMP	.2000-03

The user's input is tabulated in the preceding tables.

```
NEW PROPERTIES: CHANGED BECAUSE OF TEMPERATURE/MOISTURE EFFECTS
*
        NEW EMP
                          .3068+06
*
        NEW GMP
                           .1096+06
                           .4000+00
        NEW NUMP
        NEW CTEMP
*
                           .4797-04
                           .4547-01
*
        NEW CMPC
                           .3748+00
*
        NEW KMP
                           .1874+01
        NEW SMPT
                           .4536+04
        NEW SMPC
                           .2421+05
        NEW SMPS
                           .4669+04
        NEW BTAMP
                           .4000+00
        NEW DIFMP
                           .2000-03
```

Temperature and moisture effects on the matrix are calculated and tabulated by the computer in the preceding table. These two tables are printed out side by side.

PRIMARY COMPOSITE PROPERTIES; AS/E WITH 2% MOISTURE IN MATRIX

BASED ON INTEGRATED THEORY FOR PREDICTING THE HYGROTHERMOMECHANICAL

RESPONSE OF ADVANCED COMPOSITE STRUCTURAL COMPONENTS.

FIBER VOLUME RATIO - .550 MATRIX VOLUME RATIO - .440
VOID CONDUCTIVITY - .1000+00 VOID VOLUME RATIO - .010

1	ELASTIC MODULI	EPC1	.1776+08
2		EPC2	.9275+06
3		EPC3	.9275+06
4	SHEAR MODULI	GPC12	.5181+06
5		GPC23	.4346+06

6		GPC13	.5181+06
7	POISSON'S RATIO	NUPC12	.2860+00
8		NUPC23	.6711-01
9		NUPC13	.2860+00
10	THERM. EXP. COEF.	CTEPC1	1208-06
11		CTEPC2	.2146-04
12		CTEPC3	.2146-04
13	DENSITY	RHOPC	.5466-01
14	HEAT CAPACITY	CPC	.2450+00
15	HEAT CONDUCTIVITY	KPC1	.3198+03
16		KPC2	.5407+01
17		KPC3	.5407+01
18	STRENGTHS	SPC1T	.2220+06
19		SPC1C	.1231+06
20		SPC2T	.4470+04
21		SPC2C	.2386+05
22		SPC12	.4601+04
23	MOIST. DIFFUSIVITY	DPC1	.8800-04
24		DPC2	.5168-04
25		DPC3	.5168-04
26	MOIST. EXP. COEF.	BTAPC1	.3620-02
27		BTAPC2	.1126+00
28		BTAPC3	.1126+00
29	FLEXURAL MODULI	EPC1F	.1776+08
30		EPC2F	.9275+06
31	STRENGTHS	SPC23	.4601+04
32		SPC1F	.1980+06
33		SPC2F	.9411+04
34		SPCSB	.6902+04
35	PLY THICKNESS	TPC	.5000-02
36	INTERPLY THICKNESS	PLPC	.5850-04
37	INTERFIBER SPACING	PLPCS	.5850~04

```
FLEXURAL STRENGTH
         ******
I
         LONGITUDINAL
               I
                 TRANSVERSE
I BILINEAR
       I
          .1584+06
               I
                  .7529+04
                       Ţ
I SOLUTION
               Ŧ
I LINEAR RECTANGULAR I
         .1937+06
               I
                  .8802+04
I SOLUTION
       T
               Т
.1937+06
I NEUTRAL AXIS SHIFT I
             I
                  .8802+04
                       1
I SOLUTION
.9411+04
I PARABOLIC
          .1980+06
       1
               Ι
                       1
I SOLUTION
                       T
I RECTANGULAR
       Ι
          .2376+06
               I
                  .1129+05
                       1
I SOLUTION
```

The primary composite flexural strength table concludes the primary composite calculations. User input for the secondary composite is now tabulated.

SECONDARY FIBER PROPERTIES; S-GLASS

```
ELASTIC MODULI
                                EFS1
                                             .1240+08
                                EFS2
                                             .1240+08
 3
       SHEAR MODULI
                                GFS12
                                             .5170+07
                                GFS23
                                             .5170+07
 4
       POISSON'S RATIO
                                             .2000+00
 5
                                NUFS12
 6
                                NUFS23
                                             .2000+00
       THERM. EXP. COEF.
                                CTEFS1
                                             .2800-05
 8
                                CTEFS2
                                             .2800-05
                                             .9000--01
 9
       DENSITY
                                RHOFS
10
       NO. OF FIBERS/END
                                NFS
                                             .2040+03
       FIBER DIAMETER
                                DIFS
                                             .3600-03
11
       HEAT CAPACITY
                                CFSC
                                             .1700+00
12
                                KFS1
                                             .7500+01
13
       HEAT CONDUCTIVITY
14
                                KFS2
                                             .7500+01
15
                                KFS3
                                             .7500+01
```

16 STRENGTHS SFST .3600+06 17 SFSC .3000+06

SECONDARY MATRIX PROPERTIES; 3501-5 E WITH 2% MOISTURE

REFERENCE TEMP. - 273.00
TEST TEMP. - 294.00
MOISTURE CONDUCTIVITY - .1000+00
MOISTURE DENSITY - .1000+01
DRY GLASS TRANS. TEMP. - 460.00
PCT. MOISTURE - 2.000
MOISTURE HEAT CAPACITY - .1000+00

ORIGINAL MATRIX PROPERTIES

1	ELASTIC MODULUS	EMS	.4600+06
2	SHEAR MODULUS	GMS	.1643+06
3	POISSON'S RATIO	NUMS	.4000+00
4	THERM. EXF. COEF.	CTEMS	.3200-04
5	DENSITY	RHOMS	.4430-01
6	HEAT CAPACITY	CMSC	.2500+00
7	HEAT CONDUCTIVITY	KMS	.1250+01
8	STRENGTHS	SMST	.6800+04
9		SMSC	.3630+05
10		SMSS	.7000+04
11	MOISTURE COEF	BTAMS	.4000+00
12	DIFFUSIVITY	DIFMS	.2000-03

NEW PROPERTIES: CHANGED BECAUSE OF TEMPERATURE/MOISTURE EFFECTS

NEW EMS .3068÷06 NEW GMS .1096+06 * NEW NUMS .4000+00 * .4797-04 NEW CTEMS NEW RHOMS NEW CMSC .4547-01 .3748+00 * .1874+01 NEW KMS * .4536+04 * NEW SMST NEW SMSC .2421+05 NEW SMSS .4669+04 ×. NEW BTAMS .4000+00 * NEW DIFMS .2000-03

SECONDARY COMPOSITE PROPERTIES; S-GLASS WITH 2% MOISTURE IN MATRIX BASED ON INTEGRATED THEORY FOR PREDICTING THE HYGROTHERMOMECHANICAL RESPONSE OF ADVANCED COMPOSITE STRUCTURAL COMPONENTS.

FIBER VOLUME RATIO - .550 MATRIX VOLUME RATIO - .440 VOID CONDUCTIVITY - .1000+00 VOID VOLUME RATIO - .010

1	ELASTIC MODULI	ESC1	.6981+07
2		ESC2	.1303+07
3		ESC3	.1303+07
4	SHEAR MODULI	GSC12	.5872+06
5		GSC23	.5872+06
6		GSC13	.5872+06
7	POISSON'S RATIO	NUSC12	.2860+00
8		NUSC23	.1098+00
9		NUSC13	.2860+00
10	THERM. EXP. COEF.	CTESC1	.3840-05
11		CTESC2	.1932-04
12		CTESC3	.1932-04
13	DENSITY	RHOSC	.6951-01
14	HEAT CAPACITY	CSC	.2289+00
15	HEAT CONDUCTIVITY	KSC1	.4950+01
16		KSC2	.3617+01
17		KSC3	.3617+01

18	STRENGTHS	SSC1T	.2027+06
19		SSC1C	·9655+05
20		SSC2T	.4470+04
21		SSC2C	2386+05
22		SSC12	.4601+04
23	MOIST. DIFFUSIVITY	DSC1	.8800-04
24		DSC2	.5168-04
25		DSC3	.5168-04
26	MOIST, EXP, COEF,	BTASC1	.9209-02
27		BTASC2	.1102+00
28		BTASC3	.1102+00
29	FLEXURAL MODULI	ESC1F	.6981+07
30		ESC2F	1303+07
31	STRENGTHS	SSC23	.4601+04
32		SSC1F	.1635+06
33		SSC2F	.9411+04
34		SSCSB	.6902+04
35	PLY THICKNESS	TSC	.5000-02
36	INTERPLY THICKNESS	PLSC	7020-04
37	INTERFIBER SPACING	PLSCS	.7020-04

I***************	******	**********
I	FLEXURAL STRENGTH	1
I	**********	I
I****************	*****************	*******
2 11120111		I TRANSVERSE I
*************	*******	
I BILINEAR	.1308+06	I .7529+04 I
I SOLUTION	<u>.</u> 	**************************
	4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4	1 .8802+04 I
I LINEAR RECTANGULAR	1 .1630+06	1 10002704 1
I SOLUTION T****************	********	<u>.</u> *************
I NEUTRAL AXIS SHIFT	I .1650+06	1 .8802+04 I
I SOLUTION	T	T I
1************	*******	- *********
I PARABOLIC	I .1635+06	I •9411+04 I
I SOLUTION	I :	I
I*************	********	********
I RECTANGULAR	I •1962+06	I .1129+05 I
I SOLUTION	I :	I
I******	*********	**********

HYBRID COMPOSITE PROPERTIES; AS/E//S-G/E,80/20 WITH 2% MOISTURE IN MATRIX BASED ON PREDICTION OF PROPERTIES OF INTRAPLY HYBRID COMPOSITES (RULE OF MIXTURES).

PRIMARY COMPOSITE VOLUME RATIO - .800

SECONDARY COMPOSITE VOLUME RATIO - .200

	FLACTIC KODIU I	C1104	15/0100
1	ELASTIC MODULI	EHC1	,1560+08
2		EHC2	.1003+07
3		EHC3	.1003+07
4	SHEAR MODULI	GHC12	.5319+06
5		GHC23	.4651+06
6		GHC13	.5319+06
7	POISSON'S RATIO	NUHC12	.2860+00
8		NUHC23	.7566-01
9		NUHC13	.2860+00
10	THERM. EXP. COEF.	CTEHC1	.6714-06
11		CTEHC2	.2103-04
12		CTEHC3	.2103-04
13	DENSITY	RHOHC	.5763-01
14	HEAT CAPACITY	CHC	.2418+00
15	HEAT CONDUCTIVITY	KHC1	.2568+03
16		KHC2	.5049+01
17		KHC3	.5049+01
18	STRENGTHS	SHC1T	.2181+06
19		SHC1C	1178+06
		SHC2T	.4470+04
20		311621	177/0704

21		SHC2C	.2386+05
22		SHC12	.4601+04
23	MOIST. DIFFUSIVITY	DHC1	.8800-04
	MOTOL PILLOSIALL		
24		DHC2	.5168-04
25		DHC3	.5168-04
26	MOIST. EXP. COEF.	BTAHC1	.4738-02
27		BTAHC2	.1121+00
28		BTAHC3	.1121+00
29	FLEXURAL MODULI	EHC1F	.1560+08
30		EHC2F	.1003+07
31	STRENGTHS	SHC23	.4601+04
32		SHC1F	.1912+06
33		SHC2F	.9411+04
34		SHCSB	.6902+04
35	FIBER VOL. RATIO	VFH	.5500+00

I*******	*******	***************
I	FLEXURAL STRENGTH	I
I	**********	I
I***************	***********	k*****************
I THEORY	I LONGITUDINAL 3	I TRANSVERSE I
I ****************	************	**************
I BILINEAR	I •1530+06	[.7529+04 I
I SOLUTION	1	t į I
I ****************	**********	*************
I LINEAR RECTANGULAR	I .1882+06	I .8802+04 I
I SOLUTION	I	I
I***************	**********	************
I NEUTRAL AXIS SHIFT	I •1882+06	I .8802+04 I
I SOLUTION	I	I I
I**************	*********	************
I PARABOLIC	I .1912+06	I .9411+04 I
I SOLUTION	I :	I I
 	********	********
I RECTANGULAR	I •2295+06	I .1129+05 I
I SOLUTION	I	I
I***************	**************	***************

Using IATEST2

This method is used to check the interactive mode of INHYD. IATEST2 is a dataset that, when called, automatically inputs data as the user would enter it point by point in the interactive mode. It can be used to test this mode quickly. This INHYD.IATEST2 dataset is shown here:

```
0.20E+00
PRIMARY
FIBER
   AS GRAPHITE FIBER
32.E+06,2.E+06,2.E+06,1.E+06,.2,.25
-.56E-06,5.6E-06,.06,1.E+4,3.E-04,.17
580.0,58.0,58.0,400.E+03,400.E+03
TMAT
   EPOXY MATRIX - 3501-5 - 70F - DRY
.46E+06,.1643E+06,.4,32.E-06..443E-01,.25
1.25,6.8E+03,36.3E+03,7.E+03,0.4,2.0E-04
273.0,460.0,294.0,0.00
55/45 AS/E
.55,.45,0.0
HTM
SECONDARY
FIBER
   S-GLASS FIBER
12.4E+06,12.4E+06,5.17E+06,5.17E+06,.2,.20
2.8E-06,2.8E-06,.09,204.,3.6E-04,.17
7.5,7.5,7.5,.36E+06,.36E+06
TMAT
   EPOXY MATRIX - 3501-5 - 70F - DRY
.46E+06,.1643E+06,.4,32.E-06,.443E-01,.25
1.25,6.8E+03,36.3E+03,7.E+03,0.4,2.0E-04
273,0,460.0,294.0,0.00
55/45 S-G/E
.55,.45,0.0
HTM
80/20
        AS/E//S-G/E
RULE OF MIXTURES
```

To start a run, the user must follow this procedure, after logging on a terminal:

@ASG. A INHYD @MAP, I .ABS IN INHYD. END @XQT .ABS

The computer returns INHYD to be run in the batch or interactive (B/I) mode; the user selected interactive. Instead of entering the secondary composite volume ratio (which is in the dataset), the user enters ADD INHYD.IATEST2, as shown here.

INHYD INTERACTIVE DATA ENTRY

SECONDARY COMPOSITE VOLUME RATIO >@add inhyd.iatest2

WHICH COMPOSITE DATA TO BE ENTERED (P/S)

PROPERTIES TO BE ENTERED (C/F/M/T)

INPUT DATA HEADING (1 TO 80 CHARACTERS)

ENTER THE FIBER PROPERTY VALUES EFP2 GFP12 NUFP12 NUFP23 GFP23 EFP1 CTEFP1 CTEFP2 RHOFP NEP DIFP CEPC KFP1 KFP2 KFP3 SEPT SEPC

PROPERTIES TO BE ENTERED (C/F/M/T)

INPUT DATA HEADING (1 TO 80 CHARACTERS)

ENTER THE MATRIX PROPERTY VALUES
EMP GMP NUMP CTEMP RHOMP CMPC
KMP SMPT SMPC SMPS BTAMP DIFMP

ENTER THE THERMAL/MOISTURE PROPERTY VALUES
TO TGDR TEMP MOIST (%)

COMPOSITE HEADING (1 TO 80 CHARACTERS)

ENTER CONSTITUENT VOLUME RATIOS FIBER MATRIX VOID

METHOD FOR CALCULATING COMPOSITE PROPERTIES (H/M)

WHICH COMPOSITE DATA TO BE ENTERED (P/S)

PROPERTIES TO BE ENTERED (C/F/M/T)

INPUT DATA HEADING (1 TO 80 CHARACTERS)

ENTER THE FIBER PROPERTY VALUES
EFS1 EFS2 GFS12 GFS23 NUFS12 NUFS23
CTEFS1 CTEFS2 RHOFS NFS DIFS CFSC
KFS1 KFS2 KFS3 SFST SFSC

PROPERTIES TO BE ENTERED (C/F/M/T)

INPUT DATA HEADING (1 TO 80 CHARACTERS)

ENTER THE MATRIX PROPERTY VALUES
EMS GMS NUMS CTEMS RHOMS CMSC
KMS SMST SMSC SMSS BTAMS DIFMS

ENTER THE THERMAL/MOISTURE PROPERTY VALUES TO TGDR TEMP MOIST (%)

COMPOSITE HEADING (1 TO 80 CHARACTERS)

ENTER CONSTITUENT VOLUME RATIOS

```
FIBER MATRIX VOID

METHOD FOR CALCULATING COMPOSITE PROPERTIES (H/M)

HYBRID HEADING (1 TO 80 CHARACTERS)

METHOD FOR CALCULATING HYBRID PROPERTIES (R/M)
```

The computer then tabulates the data taken from the dataset INHYD.IATEST2. This tabulation is done in exactly the same format as is shown for the example with moisture and temperature effects used earlier and therefore is not listed here. The listing includes the tabulation of the hybrid composite properties and the flexural strengths.

INHYD Batch Mode

From Terminal

The program is started as with the interactive mode according to the system in use. At the Lewis Research Center on the UNIVAC 1110, the following procedure is used:

```
@ASG. A INHYD
@MAP, I .ABS
IN INHYD.
END
@XQT .ABS
B
```

The command selects the batch mode. Next the user adds a previously prepared dataset that contains the needed input data in the form desired. For example,

@ADD INHYD.DAT1

The format of the input data for running INHYD in a batch mode is shown on the next page for INHYD.DAT1. The "scale" entries were added to orient the input data in the proper columns and are not part of the actual data entry. Each table entry is identified after the input table.

```
>@ed;u inhed.dat1
CASE UPPER ASSUMED
ED 16R1A-FRI-04/22/83-13:59:43-(23,24)
EDIT
0:>scale
123456789012345678901234567890123456789012345678901234567890123456789012
0:>1
       0.20E+00
1:>p 20
       0.20E+00
PRIMARY
FIBER
   AS GRAPHITE FIBER
                         2.E+06
                                                         1.E+06
        32.E+06
                                         2.E+06
            .25
                       -.56E-06
                                                           .063
                                                                         1.F+04
                                        5.6E-06
         3.E-04
                            .17
                                          580.0
                                                           58.0
                                                                           58.0
       400.E+03
                       400.E+03
TMAT
   EPOXY MATRIX - 3501-5 - 70F - DRY
                     .1643E+06
                                                                       .443E-01
                                                        32.E-06
        .46E+06
                           1.25
                                       14.0E+03
                                                       36.3E+03
                                                                       16.0E+03
             .25
                        2.0E-04
            0.4
          273.0
                                          394.0
                          460.0
                                                           0.00
55/45 AS/E
                                            0.0
            .55
```

```
HTM
SECONDARY
FIBER
   S-GLASS FIBER
20:>scale
123456789012345678901234567890123456789012345678901234567890123456789012
20:>p 20
   S-GLASS FIBER
       12.4E+06
                        12.4E+06
                                        5.17E+06
                                                         5.17E+06
                                                                                . 2
        .20
3.6E-04
                         2.8E-06
                                         2.8E-06
                                                              .09
                                                                             204.
                             .17
                                              7.5
                                                              7.5
                                                                              7.5
         .36E+06
                          .3E+06
TMAT
   EPOXY MATRIX - 3501-5 - 70F - DRY
                      .1643E+06
         .46E+06
                                                                         .443E-01
                                                          32 E-06
                                        14.0E+03
             .25
                            1.25
                                                         36.3E+03
                                                                         16.0E+03
            0.4
                         2.0E-04
           273.0
                           460.0
                                           394.0
                                                             0.00
55/45 S-G/E
                             .45
             .55
                                             0.0
HTM
80/20 AS/E//S-G/E
ROM
EOF:35 SCAN:15
0:>scale
```

123456789012345678901234567890123456789012345678901234567890123456789012

The entries in the DAT1 input table (symbols defined in appendix C) are as follows:

```
PRIMARY
FIBER
   AS GRAPHITE FIBER
                           GEP12
                                      GEP23
                                                 NUFP12
    EFP1
               EFP2
    NUFP23
               CTEFP1
                           CTEFP2
                                      RHOFF
                                                 NEP
               CFPC
                           KFP1
                                      KFP2
                                                 KFP3
    DIFP
    SFPT
               SFPC
TMAT
   EPOXY MATRIX - 3501-5 - 70F -
                                     DRY
               GMP
                           NUMP
                                      CTEMP
                                                 RHOMP
    EMP
    CMPC
               KMP
                           SMPT
                                      SMPC
                                                 SMPS
    BTAMP
               DIFMP
    TO
               TGDR
                           TEMP
                                      %MOISTURE
55/45 AS/E
               MUR
                           UUR
    FVR
HTM
SECONDARY
FIBER
   S-GLASS FIBER
               EFS2
                          GFS12
                                      GFS23
                                                 NUFS12
    FFS1
               CTEFS1
                          CIEFS2
                                      RHOES
                                                 NES
    NUFS23
    DIFS
               CFSC
                          KFS1
                                      KFS2
                                                 KFS3
    SFST
               SFSC
TMAT
   EPOXY MATRIX - 3501-5 - 70F -
                                    DRY
                                                 RHOMS
               GMS
                          NUMS
                                      CTEMS
    FMS
    CMSC
               KMS
                          SMST
                                      SMSC
                                                 SMSS
    BTAMS
               DIFMS
    ΤO
               TGDR
                          TEMP
                                      %MOISTURE
55/45 S-G/E
               MUR
                          VVR
    FUR
HTM
80/20 AS/E//S-G/E
ROM
```

DAT1 contains the volume fraction of secondary composite as the first data entry; it also has the volumetric proportions of fiber, resin (matrix), and voids as well as all of the necessary fiber and matrix properties. All of these data are arranged in the proper format so that the computer can access the data and use it to calculate the primary and secondary composite properties. The program allows for temperature and moisture effects, so the subroutine HTM (which allows for hygrothermomechanical effects) is used to calculate the primary and secondary composite properties. The hybrid composite properties are then calculated by the rule of mixtures. Of course, any other property values

(and headings) desired can be substituted for those presently contained in the sample data set to make it applicable to any desired intraply hybrid composite.

DAT2 is to be used if primary composite properties are known but secondary composite properties must be calculated from fiber and matrix properties; otherwise, it is similar to DAT1 (see compiled listing). DAT3 uses fiber and matrix properties to calculate primary composite properties when secondary composite properties are known. DAT4 can be used when primary and secondary composite properties are known; it simply uses the rule of mixtures to determine hybrid composite properties. DAT5 is similar to DAT1 but with moisture present. Several other examples can be obtained by doing @ADD INHYD.DAT6, 7, 8, or 9 instead of @ADD INHYD.DAT1 as was done in the preceding example. Any other desired datasets can be made up by using the same format. Using INHYD.START results in a set of several sequential computer runs in which such things as primary and secondary composite ratios in the hybrid composite are varied or in which different properties for the matrix modulus (for example) are varied to study these effects on the calculated hybrid composite properties. To make INHYD.START run, follow this procedure:

```
@ASG. A INHYD@MAP, I .ABSIN INHYD.END@ADD INHYD.START
```

Initiated by Cards

The cards needed to initiate an INHYD run at Lewis are

```
"RUN, M/TP MSBJS, YOH1965, USERS NAME, 20, 200
"Run card includes the project number, the time, and the number of pages allowed but would vary with the requirements of the computer center.
"COL 9000FD

@ASG. A INHYD

@MAP, I .ABS
IN INHYD.
END

@XQT .ABS
B

@ADD INHYD.DAT1

It selects the specific run desired from those on file.

@COL 1100FD

"FIN
```

Concluding Remarks

This users manual provides step-by-step instructions for using the computer program INHYD (INtraply HYbrid composite Design). The instructions are given for both the user-interactive mode and the batch mode of running INHYD. The various options provided in INHYD are demonstrated with appropriate computer runs. Data input formats, sample computer outputs, and symbol lists are included. The general structure, capabilities, and options of INHYD are described in an appendix.

National Aeronautics and Space Administration Lewis Research Center Cleveland, Ohio, August 12, 1983

Appendix A Elements and Subroutines

INHYD consists of a number of elements and subroutines; they are listed alphabetically and described briefly in this section

ASD	17 properties of AS graphite in format to be used in INHYD
CHANGE 1	Edits FIBMT and COMPP to remove comment symbols (C) from three cards (lines) to activate later commands; then calls INHYD.COM when FIBMT and COMPP are used
CHANGE 2	Edits INHYD.FIBMT and COMPP, adding and removing C's from several lines; then calls INHYD.COM
CHANGE 3	Identical to CHANGE 1
CHANGE 4	Adds and removes C's before several lines of FIBMT and COMPP; then calls INHYD.COM
CHANGE 5	Adds and removes C's before several lines of FIBMT and COMPP; then calls INHYD.COM
COM	Packs and prepares INHYD; compiles FIBMT and COMPP; then initiates INHYD.START
COMPILE	Starting INHYD.COMPILE gives a compiled listing of the entire INHYD program (This depends on a UNIVAC control language program at Lewis and would not work at another computer center.)
COMPP	Generates hybrid composite properties based on micromechanics theory
DAT1	Batch run of INHYD for properties of 80/20 AS/E//3501 at 250° F. Calculations are made by using hygrothermomechanical (HTM) theory because of high-temperature use.
DAT2	Calculates hybrid composite properties for 80/20 AS/E//S-Glass/E composite. The 37 primary composite properties for a 55/45 AS graphite/epoxy resin composite are read in. The secondary composite properties are calculated by FIBMT from the S-Glass fiber and 3501 epoxy matrix properties. Then the intraply hybrid composite properties are calculated by the rule of mixtures.
DAT3	Same as DAT2 except that the secondary composite properties are read in and the primary composite is calculated by FIBMT from AS graphite and 3501 resin properties. The rule of mixtures is used to calculate hybrid composite properties.
DAT4	Calculates hybrid composite properties for 80/20 AS/E//S-Glass/E directly from the 37 primary composite and 37 secondary composite properties that are read in. The rule of mixtures is used.
DAT5	Reads in fiber and matrix properties for both primary and secondary composites. Program degrades matrix properties for the given moisture. Primary and secondary composite properties are calculated by using FIBMT (HTM can be substituted in place of FIBMT). Finally, the 80/20 AS/E//S-Glass/E hybrid composite properties are calculated from the primary and secondary composite properties by using COMPP, which uses the micromechanics theory.
DAT6	Calculates the properties of a uniaxial fibrous composite from its fiber and matrix properties by using FIBMT
DAT7	Calculates the hybrid composite properties for 80/20 HMS/E//S-Glass/E directly from the 37 primary composite and 37 secondary composite properties that are read in. The rule of mixtures is used.

DAT8 Calculates the hybrid composite properties for 80/20 AS/E//S-Glass/E by

using micromechanics theory. Input is the fiber and resin properties of the

constituents.

DAT9 Calculates the hybrid composite properties for 80/20 AS/E//S-Glass/E. Input

is the fiber and resin properties of the constituents. Primary and secondary composite properties are calculated by using micromechanics theory. Hybrid composite properties are then calculated by using the rule of mixtures. No

temperatures or moisture effects are involved.

EPOXYD Contains the 12 properties of an epoxy resin in proper format to be used by

INHYD

EPOXYMD Same as EPOXYD but with a different value for resin modulus

EXE Contains commands in INHYD and END

EXE2 Identical to EXE

EXP Contains fiber and matrix properties for a 57/43 S-Glass/3501 composite. No

moisture or temperature effects are involved.

FIBMT Calculates primary or secondary composite properties by using micro-

mechanics theory

FLEXX Calculates and prints the flexural strength of any composite by using several

theories (bilinear, linear rectangular, neutral axis shift, parabolic, and

rectangular)

HMSD Contains the 17 properties of HMS graphite fibers

HTM Calculates primary or secondary composite properties by using hygrothermo-

mechanical theory

IATEST2 Automatically inputs data as a user would enter it in interactive mode for

80/20 AS/E//S-Glass/E hybrid composite; has temperature effect; moisture

effect could be added.

IATEST3 Automatically inputs data as a user would enter it in interactive mode for

room temperature, dry, 80/20 AS/E//S-Glass/E hybrid composite

INSTR Dataset containing instructions and explanations useful to INHYD user

INTER Interactively queries user for input data required for INHYD in interactive

mode and then stores it for later use by main program

mode and then stores it is later ase of man

KEVD Contains the 17 fiber properties of Kevlar 49

LIST To get a listing of INHYD, do @START INHYD.LIST. This is not a

compiled listing (see COMPILE). (This depends on a UNIVAC control language program at Lewis and it would not work at another computer center.) This command is used for listing and compiling convenience only.

MAIN Computes intraply hybrid properties as well as primary and secondary

composite properties. Output includes elastic moduli, shear moduli, Poisson's ratios, thermal and moisture coefficients, and strengths. The equations are

based on references 2 to 4 (appendix B).

MIXRUL Predicts hybrid composite properties based on the rule of mixtures. See ref. 1

(appendix B).

PRINT This command prints the elements in the file INHYD.; also gives listings of

the following INHYD subroutines: MAIN, HTM, FIBMT, FLEXX,

MIXRUL, and COMPP.

RUN This command, used in the batch mode, starts a test run that includes

INHYD.START and INHYD.START2 (see below); it also includes several datasets identified as CHANGES (1 to 5) (see above). Use care in invoking

this command; it results in an output of about 3000 pages.

RUNTEST1 Starts INHYD.RUN from a terminal but gives listing from the computer high-

speed printer (does not complete run)

SGD Has the 17 properties of S-Glass fibers needed for determining the properties

of a S-Glass/resin composite

START Causes several sequential computer runs to be printed out of calculated prop-

erties of hybrid composites with different primary and secondary composite

ratios and composites with different fiber makeup

START2 Same as INHYD.START but uses a resin with different properties (modulus)

than START

TAPE List of INHYD elements in alphabetical order

APPENDIX B

COMPUTER CODE FOR INTRAPLY HYBRID COMPOSITE DESIGN*

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ABSTRACT

A computer program has been developed and is described herein for intraply hybrid composite design (INHYD). The program includes several composite micromechanics theories, intraply hybrid composite theories, and a hygrother-momechanical theory. These theories provide INHYD with considerable flexibility and capability that the user can exercise through several available options. Key features and capabilities of INHYD are illustrated through selected samples.

INTRODUCTION

Intraply hybrid composites generally have two kinds of fibers embedded in the matrix. They have evolved as a structural material as a logical sequel to conventional composites and to interply hybrid composites. Intraply hybrid composites have unique features that can be used to meet diverse and competing design requirements in a more cost-effective way than either advanced or conventional composites. Some of the specific advantages of intraply hybrids over the constituent composites are balanced strength and stiffness, balanced bending and membrane mechanical properties, balanced thermal distortion stability, reduced weight or cost, improved fatigue resistance, reduced notch sensitivity, improved fracture toughness or crack-arresting properties, and improved impact resistance. By using intraply hybrids, the designer can obtain a viable compromise between mechanical properties and cost to meet specified design requirements for aerospace structures.

The mechanical behavior of intraply hybrids has been investigated at the Lewis Research Center theoretically and experimentally (refs. 1 to 3). The theoretical methods and equations described in these references together with those for hygrothermal effects (ref. 4) have been integrated into a computer code for predicting hygral, thermal, and mechanical properties of, and thereby designing, intraply hybrid composites. This code is identified as INHYD for INtraply HYbrid-composite Design. The objective of this paper is to describe INHYD with respect to theory, equations, input, output, and its various options.

The capability and various options of the program are briefly described. Typical input for INHYD includes fiber and resin properties or composite properties, volume ratios of the primary and secondary fibers, glass transition temperature of the resin, moisture, and cure and use temperatures. Typical output includes moisture expansion coefficient, thermal expansion coefficients, moduli and strengths (in plane and flexural). Selected samples of

^{*}This appendix is a reprint of NASA TM-82593.

input and output are included to illustrate the flexibility and capability of INHYD. Also, the planned extensions and couplings with integrated computer programs are briefly discussed.

INTRAPLY HYBRID GEOMETRY AND DEFINITIONS

The generic geometry of intraply hybrids included in INHYD is shown schematically in figure 1. The features to be noted in this figure are as follows:

- (1) The schematic shows an eight-ply, unidirectional intraply hybrid. Each ply consists of primary composite (blank) and secondary composite (cross hatched).
- (2) The location of the secondary composite is regular and staggered through the thickness of the intraply hybrid.
- (3) The coordinate reference axes are 1, 2, and 3, where 1 is taken parallel to the fiber direction, 2 is taken transverse to the fiber direction, and 3 is taken through the thickness. These coordinate axes will be called the material axes. All properties to be discussed in subsequent sections are defined with numerical subscripts corresponding to the material axes.

It is clear from figure 1 that an intraply hybrid is made by laying tapes (tows) of primary composite and tapes (tows) of secondary composite in a definite manner in order to obtain the desired volume ratios of each. For purposes of identification, the primary composite in the hybrid is the one that constitutes the largest volume ratio. This imposes no limitations on either the capability or flexibility of INHYD. Another way to view an intraply hybrid is the one in which the constituent composites are uniformly dispersed in a typical cross-sectional area.

THEORIES INCLUDED IN INHYD

The theories programmed in INHYD have been reported previously and include (1) rule of mixtures (ref. 1), (2) micromechanics equations for hygrothermal effects (ref. 2), and (3) intraply hybrid composite micromechanics (ref. 4). Also, several equations have been programmed for predicting flexural strength and through—the—thickness shear as will be described later. These theories consist of equations that predict intraply hybrid composite properties based on constituent composite (primary and secondary) properties. The constituent composite properties may in turn be predicted by micromechanics equations using fiber and matrix properties. Essential features of these theories are summarized below. The detailed derivations, equations, and justifications are found in the original references.

The rule-of-mixtures equations have the following form (ref. 1):

$$P_{HC} = V_{PC}P_{PC} + V_{SC}P_{SC}$$
 (1)

where P denotes property and V volume ratio. The subscript HC denotes hybrid composite; PC, primary composite; and SC, secondary composite. Any property of the intraply hybrid composite can be predicted by using equation (1) when the properties of the primary and secondary composites have been specified. Equation (1), though of simple form, does satisfy the three principles of mechanics: force equilibrium, strain compatibility, and stress-strain relationships. Equation (1) predicts values that are in good agreement with measured data (ref. 1) and with other more sophisticated theories (ref. 2). Equation (1) is programmed as two subroutines in INHYD: one for

the primary composite and one for the secondary composite. These subroutines require input of constituent composite properties. The individual properties required are summarized in later sections.

The micromechanics equations from reference 4 are programmed in INHYD for predicting composite properties by using fiber and matrix properties and by accounting for hygrothermal effects. These equations are also programmed in two different subroutines: one for predicting primary composite properties and one for predicting secondary composite properties. Calls to these subroutines require inputs of fiber and matrix properties, fiber volume ratio, void volume ratio, use temperature, and moisture. The individual properties needed are described later.

The intraply hybrid composite micromechanics equations from reference 2 are programmed in INHYD. Three subroutines are used: one each for the primary and the secondary constituent composites and one that combines these two into an intraply hybrid. The first two subroutines require fiber and matrix properties; the third requires the primary and secondary composite properties predicted by the first two.

The properties predicted by all of these theories are those that result from uniform response (due to moisture, temperature, or stress) variation through the thickness of the intraply hybrid. Properties resulting from non-uniform response variation through thickness such as flexural strength and short-beam shear strength are considered in the next section.

FLEXURAL PROPERTIES

The equations programmed for flexural properties in INHYD are for flexural strength along the 1- and 2-directions and the accompanying shear strength in the 1-3 plane (short beam shear) in figure 1. Four different equations are programmed for flexural strengths. The equations have not been reported previously. The equations are readily derivable using the well-known simple beam theory and assumed stress variations through the beam thickness at fracture (fig. 3). The four equations included in INHYD were derived by assuming (1) bilinear stress distribution; (2) parabolic stress distribution; (3) linear rectangular stress variation; and (4) rectangular stress distribution (fig. 3). All these distributions are possible in intraply hybrid composites depending on their constituents. The selections of a specific equation requires user interaction based on measured data correlation. The default option in INHYD is the parabolic stress distribution. All four equations are programmed in two different subroutines: one for the primary composite and one for the secondary composite. The corresponding flexural strengths for the intraply hybrid are calculated in the same subroutine for intraply hybrid micromechanics.

The equation for the through-the-thickness shear strength in the 1-3 plane is derived by assuming parabolic shear stress variation. The integral of this parabolic variation is equated to uniform in-plane shear to obtain the desired equation. This results in the following simple relationship: The through-the-thickness shear (short-beam shear) is equal to 1.5 times the corresponding in-plane shear strength. The short-beam shear strength of the intraply hybrid is calculated in the same subroutine as the flexural strength. The equation for predicting the short-beam shear strength in the 2-3 plane (fig. 1) is similar, but it is not programmed in INHYD. The reason that it is not programmed is that experimental data indicate that specimens tested for this shear fail invariably by transverse flexure.

INHYD COMPUTER PROGRAM STRUCTURE

The logic flow of the INHYD computer program is illustrated schematically in figure 2. The theoretical functions of the program are enclosed with double lines. The input, intermediate output, and final output are enclosed with single lines. Typical final uses of the output are enclosed with dashed lines. The various micromechanics theories and requisite subroutine described previously are included in the first double-line block. Those for predicting the intraply hybrid properties are included in the second double-line block. The user controls INHYD through the main subroutine by selecting combinations of the several available options. The options in INHYD are summarized in table I. It is clear from table I that combinations of options can be specified for theory, hybrid, input property sets, and hygrothermal effects.

The input of INHYD is summarized in table II. A sample user input data set when the intraply hybrid properties will be predicted by using fiber and matrix properties is shown in table III . The first set of Booleans are the options for input read-in formats and for environmental effects. The first line of data is the secondary composite volume ratio. The second set of Booleans specify the composite micromechanics subroutines to be used. The alphanumerics describe the primary composite system. The next (third) set of Booleans define the fiber properties that will be read in. This set of Booleans is followed by the primary composite fiber properties: moduli, Poisson's ratios, thermal expansion coefficients, density, number of fibers per end, fiber diameter, heat capacity, heat conductivities, and tensile and compressive strengths. The alphanumeric card following the fiber properties describes the primary composite matrix and the environmental conditions. The next (fourth) set of Booleans specify the matrix properties to be read in. These matrix properties are moduli, moisture expansion coefficient per percent of moisture, thermal expansion coefficients, heat capacity, thermal conductivity, strengths (tensile, compressive, and shear), moisture diffusivities, glass transition temperatures, reference temperature, moisture content, and volume ratios (fiber, matrix, and void). The data following are the input data needed for the fiber and matrix in the secondary composite. The format for these data is similar to those (one for one correspondence) for the primary composite.

Formatted output (compiled data) of the input data for the primary composite fiber properties is shown in table IV and in table V for the matrix. In these tables the Booleans for the various program options, the fiber and matrix types, the volume ratios, and the constituent properties (with the corresponding program name) are displayed. The reader can readily match corresponding values from the user input data (table III) and the compiled data (tables IV and V).

INHYD program output features are summarized in table VI. A sample output for the predicted primary composite properties (using the input data tables IV and V) is illustrated in table VII. The volume ratio values and 37 properties are listed in this table. The various properties are identified by both name and assigned variable used in the program. Note the flexural properties, lines 29 to 34. Note also the ply thickness (line 35), the interply thickness (line 36), and the interfiber spacing (line 37). The format for the outputs of the secondary composite and for the intraply hybrid are similar to that for the primary (table VII).

INHYD PROGRAM CAPABILITY - ILLUSTRATED EXAMPLES

The INHYD program capability and flexibility is illustrated, in part, with typical examples. The effects of hybridization on selected properties are summarized in table VIII for 70°F, dry environmental conditions. The properties shown in this table were predicted by using the user input data in table III. Corresponding properties for the primary, the secondary, and the intraply hybrid composite are summarized. Corresponding properties for the primary composite are the same as those in table VII.

The thermal degradation, due to high temperature, in the properties of two intraply hybrids (AS/E//S-G/E and (AS/E//KEV/E) is illustrated in table IX. It is instructive to note in this table that the thermal degradation is insignificant in longitudinal properties (except longitudinal compressive strength). However, the thermal degradation is considerable (about 30 percent) for transverse and shear properties and for longitudinal compressive strength.

The hygrothermal degradation in the properties of the AS/E//S-G/E intraply hybrid is illustrated in table X. The hygrothermal environments are 70°F with 1 percent moisture and 250°F with 1 percent moisture. Again, the hygrothermal degradation is insignificant for the longitudinal properties, except for the longitudinal compressive strength. On the other hand the 1 percent moisture degrades the transverse and shear properties and the longitudinal compressive strength about 10 percent for the 70°F case and about 30 percent for the 250°F case.

INHYD EXTENSIONS AND COUPLING

The properties predicted by INHYD described previously constitute only the first part of INHYD. The program can be readily extended to predict other properties of intraply hybrids such as impact resistance and fatigue. In addition it can be made a part of or coupled with integrated programs for special or general structural analysis. The near-future planned extensions and couplings with other programs are summarized in table XI. The theory for impact resistance in reference 5 will be used. The theory for fatigue resistance is presently under development and will be reported in the ASTM 6th Conference on Composite Materials: Testing and Design, May 1981. INHYD will be coupled with an in-house laminate analysis code (MFCA, ref. 6). It will also be coupled with three integrated computer programs under in-house development: CODSTRAN - Composite Durability Structural Analysis (ref. 7); COBSTRAN - Composite Blade Structural Analysis (ref. 8); and CISTRAN - Composite Impact Structural Analysis (ref. 9).

CONCLUSIONS

A computer program called INHYD has been developed to predict the properties of unidirectional intraply hybrid composites and, therefore, to assist in the design of these hybrids. Several composite micromechanics and intraply hybrid theories and a hygrothermal mechanical theory in INHYD provide the program with considerable flexibility that the user exercises through combina-

tions of options. These options control the input data, fiber and matrix, unidirectional composite or combinations, the output, and the theory. Selected samples illustrate key features and capabilities of INHYD. INHYD provides the designer or analyst with a convenient analytical means to investigate several intraply hybrids during the preliminary design phases. INHYD can also be used to guide, and therefore keep to a minimum, required characterization of intraply hybrids.

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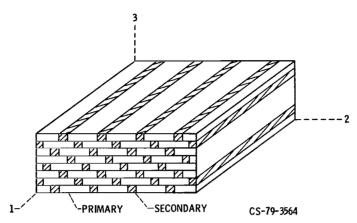


Figure 1. - Schematic of unidirectional intraply hybrid composite.

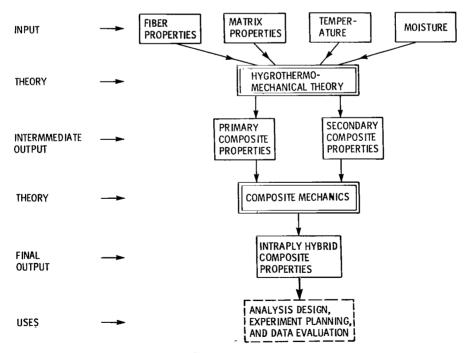


Figure 2. - Flow chart of INHYD computer program.

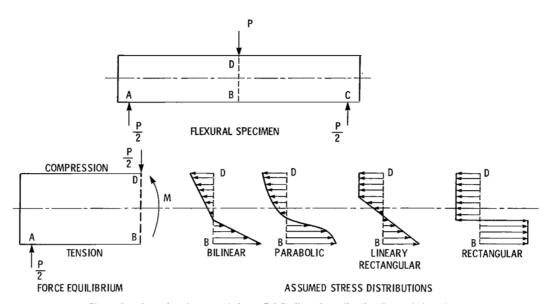


Figure 3. - Geometry, force, and stress distributions for estimating flexural strength.

TABLE I. - PROGRAM OPTIONS

Selects theory to be used

Hybrid or conventional composite

Types of property inputs

Composite properties

Constituent properties

Combinations

Hygrothermal effects

User options

Constituent properties (fiber, matrix) or properties

of primary and secondary composites or combinations

Fiber volume ratio, void volume ratio

Primary composite ratio, secondary composite ratio

Temperature, moisture

TABLE II. - PROGRAM INPUT

TABLE III. - USER INPUT DATA

```
FTTT
     0.20E+00
   Т
   T
  AS GRAPHITE FIBER
   F
    T
   F
   F
                                                    1.E+06
       32.E+06
                      2.E+06
                                     2.E+06
       .25
3.E-04
                                                      .063
                                                                     1.E+04
                     -.56E-06
                                     5.6E-06
                                       580.0
                                                       58.0
                                                                     58.0
                     .17
      400.E+03
                   400.E+03
  EPOXY MATRIX - 3501-5 - 250F - DRY
   F
   F
   F
   T
                                                                  .443E-01
       .46E+06
                    .1643E+06
                                                    32.E-06
                                                    36.3E+03
                                     6.8E+03
                                                                    7.E+03
          . 25
                        1.25
           0.4
                      2.0E-04
                        394.0
                                         0.0
         460.0
                                         0.0
          • 55
                         . 45
   F
  S-GLASS FIBER
   F
   T
   F
   F
                     12.4E+06
                                                   5.17E+06
      12.4E+06
                                   5.17E+06
                                                                         . 2
                                                                       204.
                                                    • 09
                    2.8E-06
                                    2.8E-06
          .20
 3.6E-04 .17
.36E+06 .3E+06
EPOXY MATRIX - 3501-5 - 250F - DRY
                                        7.5
                                                        7.5
                                                                       7.5
```

TABLE IV. - COMPILED DATA - FIBER

MRUL PRIM COMP PRIM	= T		τ	SHTMF= Mat=	T F	TMAT=	F
	2 3 SHE 4 5 POI 6 7 THE 8 9 DEN 10 NOI 11 FIE 12 HE 13 HE 14	ASTIC MODULI ESSON'S RATERM. EXP. COMMENTER OF FIBERS	10 0EF. /END R	EFP1 EFP2 GFP23 NUFP12 NUFP23 CTEFP2 RHOFP NFP DIFP CFPC KFP1 KFP2 KFP3 SFPT		3200+08 2000+07 2000+07 1000+07 2000+00 2500+00 5600-05 6300-03 1700+00 5800+03 5800+02 4000+06	
		ABLE V COM				-	
COMP= PRIMARY MA EPOXY MATRIX	ATRIX PROPE			: F	TMAT=	Ť	
TGDR=	.4600+03	Ţ=	=	.3940+03		H =	.0000
	ORIGINAL	MATRIX PRO	PERTIE	S			
1 2 3 4 5 6 7 8 9 10 11	DENSITY HEAT CAP	DULUS S RATIO XP. COEF. ACITY DUCTIVITY S COEF.	CT RH CM KM SM SM BT	P MP EMP OMP PC	.4600+0 .1643+0 .4000+0 .3200-0 .2500+0 .1250+0 .6800+0 .3630+0 .7000+0 .4000+0	06 00 04 01 00 01 04 05 04	

TABLE VI. - OUTPUT

- PRIMARY AND SECONDARY COMPOSITE PROPERTIES
 - 3 EACH, ELASTIC MODULI, SHEAR MODULI, POISSON'S RATIOS, THERM. EXP. COEF. DENSITY, HEAT CAPACITY
 - 3 HEAT CONDUCTIVITIES, 5 IN-PLANE STRENGTHS
 - 3 EACH, MOISTURE DIFFUSIVITIES AND MOISTURE EXPANSION COEFS.
 - 2 FLEXURAL MODULI
 - 4 THROUGH-THE-THICKNESS STRENGTHS
 - PLY THICKNESS, INTERPLY THICKNESS, INTERFIBER SPACING
- INTRAPLY HYBRID COMPOSITE PROPERTIES

SAME AS ABOVE EXCEPT FOR LAST LINE PLUS COMPOSITE FIBER VOLUME RATIO

TABLE VII. - TYPICAL COMPUTER OUTPUT OF INTRAPLY HYBRID PROPERTIES

PRIMARY COMPOSITE PROPERTIES

VFP=	.550	0+00 VMF=	.4500+00	VVP=	.0000
	1	ELASTIC MODULI	EPC1	.1780+08	
	2		EPC2	.1034+07	
	3		EFC3	.1034+07	
	4	SHEAR MODULI	GP:C12	.4902+06	
	5		GPC23	.4148+06	
	6		GFC13	.4902+06	
	7	POISSON'S RATIO	NUFC12	.2900+00	
	8		NUFC23	.2465+00	
	9		NUFC13	.2900+00	
	10	THERM. EXP. COEF.	CTEFC1	1816-06	
	11		CTEPC2	.1640-04	
	12		CTEPC3	.1640-04	
	13	DENSITY	RHOPC	.5458-01	
	14	HEAT CAPACITY	CPC	.2048+00	
	15	HEAT CONDUCTIVITY	KF:C1	.3196+03	
	16		KPC2	.3916+01	
	17		NPC3	.3916+01	
	18	STRENGTHS	SPC1T	.2224+06	
	19		SF'C1C	.1712+06	
	20		SPC2T	.6407+04	
	21		SFC2C	.3420+05	
	22		SFC12	.6595+04	
	23	MOIST. DIFFUSIVITY	r DPC1	.9000-04	
	24		DPC2	.5168-04	
	25		DF:C3	.5168-04	
	26	MOIST. EXP. COEF.	BTAPC1	.4384-02	
	27		BTAPC2	.1131+00	
	28		BTAPC3	.1131+00	
	29	FLEXURAL MODULI	EPC1F	.1780+08	
	30		EPC2F	.1034+07	
	31	STRENGTHS	SFC23	.6595+04	
	32		SPC1F	.2418+06	
	33		SPC2F	.1349+05	
	34		SPCSB	.9893+04	
	35	PLY THICKNESS	TPC	.5000-02	
	36	INTERFLY THICKNESS	6 PLPC	.5850-04	
	37	INTERFIBER SPACING	FLPCS	.5850-04	

TABLE VIII. - EFFECT OF HYBRIDIZATION ON COMPOSITE PROPERTIES

[Dry; temperature, 70° F.]

Property	AS/E primary composite, 80 percent	S-G/E secondary composite, 20 percent	Intraply hybrid composite, 80/20 AS/E//S-G/E
Elastic moduli: EHC1 EHC2	0.1780+08 0.1034+07	0.7015+07 0.1524+07	0.1564+08 0.1132+07
Shear moduli: GHC12 GHC13	0.4902+06 0.4148+06	0.5517+06 0.5517+06	0.5025+06 0.4422+06
Poisson's ratio, NUHC12	0.2900+00	0.2900+00	0.2900+00
Strengths: SHC1T SHC1C SHC2T SHC2C SHC2C SHC12	0.2224+06 0.1712+06 0.6407+04 0.3420+05 0.6595+04	0.2037+06 0.1448+06 0.6407+04 0.3420+05 0.6595+04	0.2187+06 0.1659+06 0.6407+04 0.3420+05 0.6595+04
Flexural moduli: EHC1F EHC2F	0.1780+08 0.1034+07	0.7015+07 0.1524+07	0.1564+08 0.1132+07

TABLE IX. - ILLUSTRATION OF THERMAL DEGRADATION IN INTRAPLY HYBRIDS

Intraply hybrid com- posite property	80/20 AS/	E//S-G/E	80/20 AS/E//KEV/E			
posite property	Temperature, °F					
	70	250	70	250		
Elastic moduli: EHC1 EHC2	0.1564+08 0.1132+07	0.1557+08 0.8067+06	0.1670+08 0.9365+06	0.1662+08 0.6995+06		
Shear moduli: GHC12 GHC23	0.5025+06 0.4422+06	0.3368+06 0.3077+06	0.4504+06 0.3715+06	0.3104+06 0.2693+06		
Poisson's ratio, NUHC12	0.2900+00	0.2900+00	0.3065+00	0.3065+00		
Strengths: SHC1T SHC1C SHC2T SHC2C SHC2C SHC12	0.2187+06 0.1659+06 0.6407+04 0.3420+05 0.6595+04	0.2175+06 0.1432+06 0.4040+04 0.2157+05 0.4159+04	0.2227+06 0.1531+06 0.6407+04 0.3420+05 0.6595+04	0.2217+06 0.1306+06 0.4040+04 0.2157+05 0.4159+04		
Flexural moduli: EHC1F EHC2F	0.1564+08 0.1132+07	0.1557+08 0.8067+06	0.1670+08 0.9365+06	0.1662+08 0.6995+06		

TABLE X. - ILLUSTRATION OF HYGROTHERMAL DEGRADATION IN INTRAPLY HYBRID (80/20 AS/E//S-G/E)

1111011 21 11151125 (00) 20 110, 21, 70 0, 21,					
Intraply hybrid com- posite property	Temperature, °F				
posite property		70	250		
	Dry	1.0 Percent moisture	Dry	1.0 Percent moisture	
Elastic moduli: EHC1 EHC2	0.1564+08 0.1132+07	0.1562+08 0.1033+07	0.1557+08 0.8067+06	0.1552+08 0.5713+06	
Shear moduli: GHC12 GHC23	0.5025+06 0.4422+06	0.4500+06 0.4006+06	0.3368+06 0.3077+06	0.2279+06 0.2140+06	
Poisson's ratio, NUHC12	0.2900+00	0.2900+00	0.2900+00	0.2900+00	
Strengths: SHC1T SHC1C SHC2T SHC2C SHC12	0.2187+06 0.1659+06 0.6407+04 0.3420+05 0.6595+04	0.2183+06 0.1584+06 0.5626+04 0.3003+05 0.5791+04	0.2175+06 0.1432+06 0.4040+04 0.2157+05 0.4159+04	0.2169+06 0.1297+06 0.2631+04 0.1405+05 0.2709+04	
Flexural moduli: EHC1F EHC2F	0.1564+08 0.1132+07	0.1562+08 0.1033+07	0.1557+08 0.8067+06	0.1552+08 0.5713+06	

TABLE XI. - PLANNED EXTENSION

- PROGRAM AND DOCUMENTATION WILL BE MADE AVAILABLE TO THE PUBLIC THROUGH COSMIC
- EXTEND PROGRAM TO DETERMINE IMPACT RESISTANCE, DEFECT PROPAGATION, AND FATIGUE RESISTANCE OF UNIDIRECTIONAL INTRAPLY HYBRIDS
- COUPLE WITH MULTILAYERED FILAMENTARY LAMINATE ANALYSIS (MFCA)
- · COUPLE INHYD WITH
 - CODSTRAN COMPOSITE DURABILITY STRUCTURAL ANALYSIS
 - COBSTRAN COMPOSITE BLADE STRUCTURAL ANALYSIS
 - CISTRAN COMPOSITE IMPACT STRUCTURAL ANALYSIS
- CONVERT PROGRAM TO IBM 370/3033

Appendix C Symbols

This appendix defines the symbols used in the computer program INHYD. Part I displays and defines the property symbols as they appear in INHYD. Part II lists in alphabetical order additional symbols that are used to represent subscripts. In the program the symbols from parts I and II are used together to further define the property.

INHYD. Property Symbols

Primary fiber properties:

EFP1,EFP2 elastic moduli
GFP12,GFP23 shear moduli
NUFP12,NUFP23 Poisson's ratios

CTEFP1, CTEFP2 thermal expansion coefficients

RHOFP density

NFP number of fibers per end

DIFP fiber diameter
CFPC heat capacity
KFP1,KFP2,KFP3 heat conductivities

SFPT,SFPC strengths

Primary matrix properties:

EMP elastic modulus
GMP shear modulus
NUMP Poisson's ratio

CTEMP thermal expansion coefficient

RHOMP density

CMPC heat capacity
KMP heat conductivity

SMPT,SMPC,SMPS strengths

BTAMP moisture coefficient

DIFMP diffusivity

Primary composite properties:

EPC1,EPC2,EPC3 elastic moduli GPC12,GPC23,GPC13 shear moduli NUPC12,NUPC23,NUPC13 Poisson's ratios

CTEPC1,CTEPC2,CTEPC3 thermal expansion coefficients

RHOPC density

CPC heat capacity

KPC1,KPC2,KPC3 heat conductivities

SPC1T,SPC1C, strengths

SPC2T,SPC2C,

SPC12

DPC1,DPC2,DPC3 moisture diffusivities

BTAPC1,BTAPC2,BTAPC3 moisture expansion coefficients

EPC1F,EPC2F flexural moduli

SPC23,SPC1F strengths

SPC2F,SPCSB

TPC ply thickness
PLPC interply thickness
PLPCS interfiber spacing

Secondary fiber properties:

EFS1,EFS2 elastic moduli GFS12,GFS23 shear moduli NUFS12,NUFS23 Poisson's ratios

CTEFS1, CTEFS2 thermal expansion coefficients

RHOFS density

NFS number of fibers per end

DIFS fiber diameter
CFSC heat capacity
KFS1,KFS2,KFS3 heat conductivities

SFST,SFSC strengths

Secondary matrix properties:

EMS elastic modulus
GMS shear modulus
NUMS Poisson's ratio

CTEMS thermal expansion coefficient

RHOMS density

CMSC heat capacity
KMS heat conductivity

SMST,SMSC,SMSS strengths

BTAMS moisture coefficient

DIFMS diffusivity

Secondary composite properties:

ESC1,ESC2,ESC3 elastic moduli GSC12,GSC23,GSC13 shear moduli NUSC12,NUSC23,NUSC13 Poisson's ratio

CTESC1, CTESC2, CTESC3 thermal expansion coefficients

RHOSC density

CSC heat capacity
KSC1,KSC2,KSC3 heat conductivities

SSC1T,SSC1C,

strengths

SSC2T, SSC2C,

SSC12

DSC1,DSC2,DSC3 moisture diffusivities

BTASC1,BTASC2,BTASC3 moisture expansion coefficients

ESC1F,ESC2F flexural moduli

SSC23,SSC1F, strengths

SSC2F,SSCSB

TSC ply thickness
PLSC interply thickness
PLSCS interfiber spacing

Hybrid composite properties:

EHC1,EHC2,EHC3 elastic moduli GHC12,GHC23,GHC13 shear moduli NUHC12,NUHC23,NUHC13 Poisson's ratio

CTEHC1, CTEHC2, CTEHC3 thermal expansion coefficients

RHOHC density

CHC heat capacity

KHC1,KHC2,KHC3 heat conductivities

SHC1T,SHC1C, strengths

SHC2T,SHC2C,

SHC12

DHC1,DHC2,DHC3 moisture diffusivities

BTAHC1,BTAHC2,BTAHC3 moisture expansion coefficients

EHC1F,EHC2F flexural moduli

SHC23,SHC1F, strengths

SHC2F,SHCSB

VFH fiber volume ratio

Primary matrix properties with moisture:

TO reference temperature
TEMP test temperature
KMST moisture conductivity

RHOMST moisture density

TGDR dry glass transition temperature MOIST% moisture content, percent

CMST moisture heat capacity

INHYD. Subscripts

B bending

C composite; compression

F fiber; flexural

H hybrid P primary

S secondary; shear

1 material axes parallel to fiber direction
2 material axes transverse to fiber direction

3 material axes through thickness

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